SRB

STS-103 (BI099) FLIGHT READINESS REVIEW

Program

November 19, 1999

SOLID ROCKET BOOSTER



AGENDA

Presenter:
Roger Elliott
Organization/Date:
USA/11-19-99

- STS-93 Postflight Assessment
 - RH Tilt Hydraulic Pressure Measurement IFA
- Certification Status No Issues
- Configuration Summary
 - Four RH EDAS and LH/RH ET Camera Configuration
- Special Topics
 - BSM Liner Soft Spot
 - Hydraulic Pump Torsion Spring
- Readiness Assessment





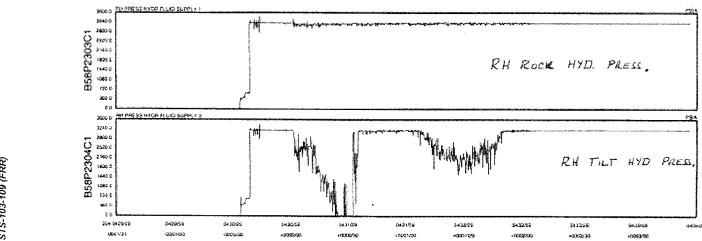
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Anomaly

 RH TVC tilt hydraulic pressure measurement became erratic during ascent

Discussion

- Pressure measurement provides data during T-17 seconds to T-10 seconds for launch commit
- Worst case impact is launch scrub (Criticality 3)

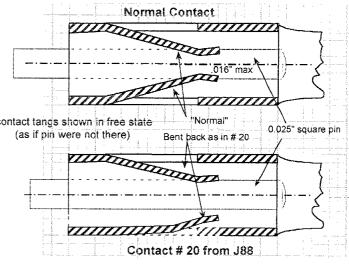


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Investigation

- Failure was not repeated
 - Pressure transducer vibrated at flight levels (sixteen additional missions)
 - Watertight reusable cable X13W3R visually inspected and electrically tested
 - Aft skirt throwaway cable X02W11
 visually inspected, electrically tested
 and subjected to destructive failure analysis
 - Aft IEA S/N 61 visually inspected, recertification tested, and vibration tested
- MDM static testing complete (no anomalies)
- Amplifier Buffer Attenuator (ABA) card testing completed (no anomalies)
- Internal inspection of IEA found one ABA card edge connector contact damaged



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Investigation (cont.)

- Intermittent open circuit on this contact would cause observed anomalous condition (failure mode validated with circuit analysis)
- Isolated case 1,485 contacts inspected
- This type of contact only used in Criticality 3 circuits
- IFA to be closed as an unexplained anomaly; most probable cause is damaged ABA connector contact

Corrective Actions

- Improved connector/contact visual inspection for these IEA signal conditioner connectors
- Visual inspection to be performed prior to reinstalling replaceable units into these connectors each time





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Rationale for Flight

- Isolated case (first occurrence)
- This connector contact type is restricted to data circuits (Criticality 3 measurements)
- No effect on TVC system control during flight
- Not a flight safety concern



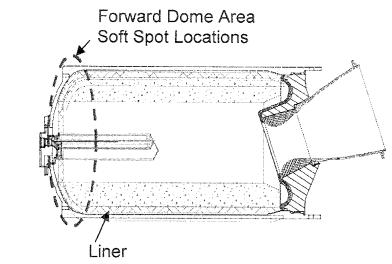
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Owner and the Company	Roger Elliott	***************************************
	Organization/Date:	-
	USA/11-19-99	***************************************

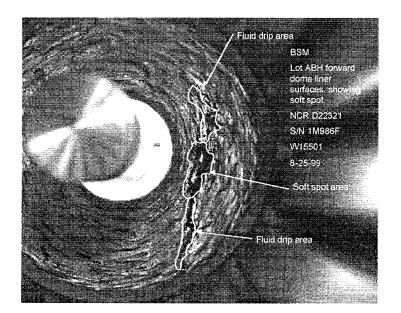
<u>Issue</u>

 Liner soft spots observed in the BSM forward dome and case wall liners during production of Lots ABG, ABH and ABJ

Concern

 Forward dome burn through and/or propellant debris









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Discussion/Background

- Total of 180 BSMs in Lots ABG, ABH and ABJ
- Seven motors had one soft spot each
 - Lot ABG (1); Lot ABH (2); Lot ABJ (4)
 - One Lot ABH BSM propellant machined out of case and one soft spot identified in liner on case wall
- No history of anomaly prior to Lot ABG
- BSMs installed on STS-103 from different lots (ABC, ABD and ABE)





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Investigation

- Investigation and testing identified root cause of soft liner to be the liner mixer in Station 250
 - Mixer rotation not in correct direction
 - Wiper blade is installed upside down not properly wiping the bowl wall
 - Mixer was only used by CSD for liner Lots ABG, ABH and ABJ
 - Uncured liner material taken from mixer side wall shows Lots ABG, ABH and ABJ produced material below specification
- BSM production Lots ABF and earlier were produced with Station 20 mixer





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Corrective Action

- Detailed review and update of liner process floor paper conducted
- Lots ABG, ABH and ABJ have been placed in bonded storage

Rationale for Flight

 STS-103 BSMs not affected; BSMs processed with different mixer, which was properly configured, and produced to specification liner material



SPECIAL TOPIC - HYDRAULIC PUMP TORSION SPRING

Presenter: Roger Elliott

Organization/Date: USA/11-19-99

<u>Issue</u>

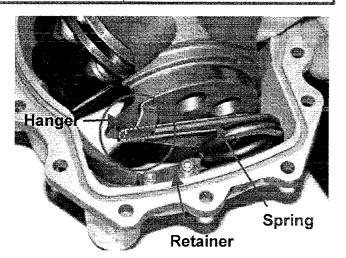
 Orbiter found Hydraulic Pump torsion spring not properly seated in hanger

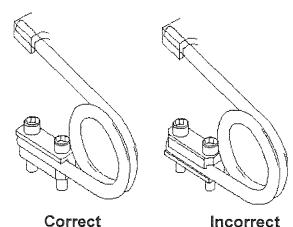
Concern

 Could cause erratic discharge pressure or loss of hydraulic system function

Discussion

 Orbiter pump flew in this configuration on STS-89 (pump functioned nominally)





Retainer Configurations

SPECIAL TOPIC - HYDRAULIC PUMP TORSION SPRING

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Discussion (cont.)

- Radiographic Inspection (RT) found to be excellent method to determine if springs are in hangers
- SRB found one pump in stores with both springs out of hangers
- Analysis shows 1900+ g's required to lift spring out of hanger
- RT of STS-103 orbiter and SRB hydraulic pumps performed (springs installed in hangers)
- Retainers used to secure torsion spring to pump housing not installed per drawing
 - Vendor testing verifies pump performance nominal with retainer installed in either configuration





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Corrective Action

- Verification techniques of proper installation after vendor refurbishment under evaluation
- Vendor engineering change in work to allow observed retainer configuration as alternate

Rationale for Flight

- RT verification of proper torsion spring/hanger installation
- Analysis shows spring will stay captured in hanger and can not "jump" out
- Either orientation of spring retainer is acceptable
- No impact to system function or flight safety





READINESS ASSESSMENT

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Pending completion of open work, there are no constraints for flight for STS-103